

**VOLUME I
PERFORMANCE FLIGHT TESTING**

**APPENDIX A
SYMBOLS
TERMS AND ABBREVIATIONS**

1982

DTIC QUALITY INSPECTED 4

19970116 075

**USAF TEST PILOT SCHOOL
EDWARDS AFB, CA**

DISTRIBUTION STATEMENT A

Approved for public release;
Distribution Unlimited

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|---------------------------|----------------------------------|
| a | Acceleration | ft/sec ² |
| a | Lift curve slope | per deg or per rad |
| a | Speed of sound | ft/sec, mi/hr, kts |
| ac | Aerodynamic center | |
| A | Area | ft ² , m ² |
| AR | Aspect ratio | |
| b | Wingspan | ft, m |
| | Blade Width | ft, m |
| B | Number of blades | |
| BHP | Brake horsepower | |
| B.L. | Base line | |
| c | Absolute velocity | |
| c | Chord | ft, m |
| C | Compression | |
| C | Specific fuel consumption | lb/hr |
| °C | Degrees centigrade | deg |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|------------------------------------|---------------------|
| C_r | Root chord | ft, m |
| C_t | Tip chord | ft, m |
| C_p | Specific heat at constant pressure | btu/lb $^{\circ}$ R |
| C_v | Specific heat at constant volume | btu/lb $^{\circ}$ R |
| C_d | Section drag coefficient | |
| C_f | Skin friction coefficient | |
| C_l | Section lift coefficient | |
| C_m | Section moment coefficient | |
| C_F | Force coefficient | |
| C_D | Aircraft drag coefficient | |
| C_L | Aircraft lift coefficient | |
| $C_{L_{ic}}$ | Indicated lift coefficient | |
| C_M | Aircraft moment coefficient | |
| C_p | Pressure coefficient | |

SYMBOLS, TERMS, AND ABBREVIATIONS

| ARABIC <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|---------------------------------|------------------------------|--------------------|
| C_P | Propeller power coefficient | |
| C_Q | Propeller torque coefficient | |
| C_T | Propeller thrust coefficient | |
| cg | Center of gravity | |
| cp | Center of pressure | |
| CR | Compression ratio | |
| CPR | Compressor Pressure ratio | |
| d | Differential | |
| D | Diameter | ft |
| D | Drag | lb |
| D | Diffuser | |
| d/dt | Time rate of change | |
| $dC_L/d\alpha$ | Lift curve slope | per deg or per rad |
| e | Oswald's efficiency factor | |
| E | Shear modulus | |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|-----------------------------|--------------|
| E | Endurance | hr |
| E | Total energy | ft lbs |
| E_m | Maneuver energy | ft lbs |
| E_s | Specific energy | ft |
| EGT | Exhaust gas temperature | deg |
| f | Function of | |
| f | Equivalent flat plate area | ft^2 |
| F | Force | lb |
| F | Fan | |
| F | Resultant aerodynamic force | lb |
| $^{\circ}F$ | Degrees Fahrenheit | deg |
| F_g | Gross thrust | lb |
| F_n | Net thrust | lb |
| F_{ex} | Excess thrust | lb |
| F.R.L | Fuselage reference line | |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|-----------------------------|--|
| F.S. | Fuselage station | |
| g | Acceleration due to gravity | ft/sec ² |
| G | Gravitational constant | 32.17405 ft ² /sec ² geo- potential ft |
| h | Enthalpy | btu/lb |
| h | Tapeline altitude | ft |
| h _v | Kinetic energy | |
| H | Total head pressure | lb/in ² |
| H | Combustor | |
| H | Altitude, general | ft |
| H | Geopotential at a point | geopotential ft |
| H _c | Pressure altitude | ft |
| H _i | Indicated altitude | ft |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC</u> <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|--|---|--------------|
| H_{ic} | Indicated altitude corrected for instrument error, $H_{ic} + \Delta H_{ic}$ | ft |
| ΔH_{ic} | Altimeter instrument correction | ft |
| H_{ic_l} | Indicated altitude corrected for instrument and lag errors, $H_i + \Delta H_{ic} + \Delta H_{ic_l}$ | ft |
| ΔH_{ic_l} | Altimeter lag correction | ft |
| ΔH_p | Altimeter position error corresponding to ΔP_p | ft |
| ΔH_{pc} | Altimeter position error correction | ft |
| HP | Horsepower | hp |
| H.V. | Heating value of hydrocarbon fuel | btu/lb |
| I_s | Specific impulse | sec |
| J | Propeller advance ratio | |
| K_n | A constant | |
| K_t | Temperature probe recovery factor | |
| $^{\circ}K$ | Degrees Kelvin | deg |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|---|--------------|
| KE | Kinetic Energy | |
| l | Characteristic length | ft |
| \ln | Natural logarithm | |
| L | Lift | lb |
| L | Length, dimensional analysis | |
| L | Standard lapse rate -1.98 °C/1000 ft | deg/ft |
| m | Slope of a line at a point | |
| m | Mass | slug |
| mac | Mean aerodynamic chord | |
| M | Mass, dimensional analysis | |
| M | Mach, flight or free stream | |
| M_i | Indicated Mach | |
| M_{ic} | Indicated Mach corrected for instrument error, $M_i + \Delta M_{ic}$ | |
| ΔM_{ic} | Machmeter instrument correction | |
| ΔM_p | Machmeter position error corresponding to P_p | |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC</u> <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|--|---|--------------------|
| ΔM_{pc} | Machmeter position error correction | |
| M | Moment | ft lb |
| MAC | Mean aerodynamic chord | |
| n | Load factor | |
| n | Number of stages | |
| N | Nozzle | |
| N | Revolutions per minute | |
| NACA | National Advisory Committee for Aeronautics | |
| NASA | National Aeronautics and Space Administration | |
| N_{pr} | Prandtl number | |
| P | Power | hp, ft lb/sec |
| P | Pressure, general | lb/in ² |
| P | The applied pressure at a point at a time, t | in Hg |
| P_a | Atmospheric pressure corresponding to H_c | in Hg |

SYMBOLS, TERMS, AND ABBREVIATIONS

| ARABIC <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|---------------------------------|--|--|
| $P_{a_{sl}}$ | Atmospheric pressure at standard sea level | 2116.22 lb/ft ² 29.92126 in Hg |
| P_i | The indicated pressure at a point at a time, t | in Hg |
| ΔP_p | Static pressure error or position error | in Hg |
| P_s | Pressure corresponding to H_{ic} | in Hg |
| P_s | Specific Excess power | |
| P_t or P_T | Free stream total pressure | in Hg, lb/in ² |
| P'_t | Total pressure at total pressure source | in Hg |
| PE | Potential energy | ft lb |
| q | Dynamic pressure, $\rho V_T^2/2$ | in Hg |
| q_c | Differential pressure, $P'_t - P_a$ | in Hg |
| q_{cic} | Differential pressure corresponding to V_{ic} , $P'_t - P_s$ | in Hg |
| Q | Heat or heat energy | btu |
| Q | Torque | in lb |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|---------------------------------|--------------------------------------|
| r | Blade length | in, ft |
| R | Radius of turn | |
| R | Range | |
| R | Gas constant for dry air | ft ² /sec ² °R |
| °R | Degrees Rankine | deg |
| R _e | Radius of the earth | ft |
| R _e | Reynolds Number | |
| RF | Range factor | |
| ROC | Required operational capability | |
| ROC | Rate of climb | |
| RW | Relative wind | |
| s | Specific Entropy | btu/lb |
| s | Distance | ft |
| S | Total wing or planform area | ft ² |
| S _a | Air distance | ft |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|--|----------------------|
| S_g | Ground roll distance | ft |
| SM | Stall margin | |
| SR | Specific range | nam |
| SFC | Specific fuel consumption | |
| SPR | Stage pressure ratio | |
| t | Thickness | in, ft |
| t | Time | sec |
| t_a | Atmospheric temperature | $^{\circ}\text{C}$ |
| t_{as} | Standard day atmospheric temperature corresponding to H_c | $^{\circ}\text{C}$ |
| t_{asl} | Standard sea level atmospheric temperature | 15°C |
| t_{at} | Test day atmospheric temperature | $^{\circ}\text{C}$ |
| t_i | Indicated temperature | $^{\circ}\text{C}$ |
| t_{ic} | Indicated temperature corrected for instrument error, $t_i + \Delta t_{ic}$ | $^{\circ}\text{C}$ |
| Δt_{ic} | Air temperature instrument correction | $^{\circ}\text{C}$ |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|---|--------------------------|
| T | Temperature | deg |
| T | Time, dimensional analysis | |
| T | Turbine | |
| T | Propeller thrust | lb |
| T_a | Atmospheric temperature | $^{\circ}\text{K}$ |
| T_{as} | Standard day atmospheric temperature corresponding to H_c | $^{\circ}\text{K}$ |
| T_{asl} | Standard sea level atmospheric temperature | 288.16°K |
| T_{at} | Test day atmospheric temperature | $^{\circ}\text{K}$ |
| T_i | Indicated temperature | $^{\circ}\text{K}$ |
| T_{ic} | Indicated temperature corrected for instrument error, $T_i + \Delta T_{ic}$ | $^{\circ}\text{K}$ |
| ΔT_{ic} | Air temperature instrument correction | $^{\circ}\text{K}$ |
| T_t | Total temperature | $^{\circ}\text{K}$ |
| T_T | Total temperature (general) | deg |

SYMBOLS, TERMS, AND ABBREVIATIONS

| ARABIC <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|---------------------------------|---|--------------|
| TE | Total energy | |
| THP | Thrust horsepower | |
| TIT | Turbine inlet temperature | deg |
| TPR | Total pressure ratio | |
| TSFC | Thrust specific fuel consumption | lb/hr |
| u | Linear velocity | ft/sec |
| V | Velocity or true airspeed | |
| V_c | Calibrated airspeed, $V_i + \Delta V_{ic} + \Delta V_{pc}$ | kts |
| V_e | Equivalent airspeed, $V_c + \Delta V_c$ or $V\sqrt{\sigma}$ | kts |
| V_i | Indicated airspeed | kts |
| V_{ic} | Indicated airspeed corrected for instrument error, $V_i + \Delta V_{ic}$ | kts |
| ΔV_{ic} | Airspeed indicator instrument correction | kts |
| V_{ic_l} | Indicated airspeed corrected for instrument and lag errors, $V_i + \Delta V_{ic} + \Delta V_{ic}$ | kts |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>ARABIC Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|----------------------------------|--|-----------------|
| ΔV_{ic_l} | Airspeed indicator lag corrections | kts |
| ΔV_p | Airspeed indicator position error corresponding to ΔP_p | kts |
| ΔV_{pc} | Airspeed indicator position error correction | kts |
| ΔV_c | Compressibility correction | kts |
| V_s | Standard day true airspeed | kts |
| V_t | Test day true airspeed | kts |
| w | Relative velocity | ft/sec |
| w or W | Work | ft/lb |
| w | Downwash velocity | ft/sec |
| W | Aircraft gross weight | lb |
| \dot{w}_a | Airflow rate | lb/hr or lb/sec |
| \dot{w}_f | Fuel flow rate | lb/hr or lb/sec |
| W.L. | Water line | |
| x | Distance | ft |
| z | Energy reference height | ft |
| \propto | Proportional to | |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|-----------------------|------------------------------------|--------------|
| α | Angle of attack | deg, rad |
| β | Angle of sideslip | deg |
| β | Bypass ratio | |
| γ | Ratio of specific heats | |
| γ | Flight path angle | deg |
| δ | Pressure ratio , $P_a/P_{a_{sl}}$ | |
| δ_{ic} | $P_s/P_{a_{sl}}$ | |
| δ_L | Laminar boundary layer thickness | |
| δ_T | Turbulent boundary layer thickness | |
| δ | Wedge angle or turning angle | |
| Δ | Change in any quantity | |
| ϵ | Axial strain | |
| ϵ | Downwash angle | deg, rad |
| η | Efficiency | |
| η_o | Overall efficiency | |
| η_p | Propulsive efficiency | |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|-----------------------|--|--------------|
| η_{th} | Thermal efficiency | |
| θ | Temperature ratio, $T_a/T_{a_{sl}}$ | |
| θ_s | $T_{as}/T_{a_{sl}}$ | |
| θ_t | $T_{at}/T_{a_{sl}}$ | |
| θ | Shock wave angle | |
| λ | Lag constant | sec |
| $\lambda_{H_{ic}}$ | Lag constant corresponding to H_{ic} | sec |
| λ_s | Static pressure lag constant | sec |
| λ_{sl} | Lag constant at standard sea level | sec |
| $\lambda_{s_{sl}}$ | Static pressure lag constant at standard sea level | sec |
| λ_t | Total pressure lag constant | sec |
| $\lambda_{t_{sl}}$ | Total pressure lag constant at standard sea level | sec |
| λ | Taper ratio | |
| λ | Sweep angle | deg |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|-----------------------|--|---|
| μ | Coefficient of absolute viscosity | lb sec/ft ² |
| μ | Viscosity at temperature T_a | lb sec/ft ² |
| $\mu_{H_{ic}}$ | Viscosity corresponding to H_{ic} | lb sec/ft ² |
| μ_{sl} | Viscosity at standard sea level | 3.7452×10^{-7} lb sec/ft ² |
| μ | Mach angle | deg |
| μ | Coefficient of friction | |
| ν | Kinematic viscosity | ft sec |
| ν | Turning angle | deg |
| π | 3.14159 ... | |
| π | Buckingham π | |
| ρ | Air density | slug/ft ³ |
| ρ_a | Standard day air density corresponding to H_c | slug/ft ³ |
| ρ_{sl} | Air density at standard sea level | .0023769 slug/ft ³ |
| ρ_t | Test day air density | slug/ft ³ |
| σ | Density ratio, ρ_a/ρ_{sl} | |
| σ_s | ρ_s/ρ_{sl} | |

SYMBOLS, TERMS, AND ABBREVIATIONS

| <u>Symbol or Term</u> | <u>Definition</u> | <u>Units</u> |
|-----------------------|--------------------|--------------------|
| σ_t | ρ_t/ρ_{sl} | |
| σ | Axial stress | lb/in ² |
| σ | Solidity ratio | |
| τ | Acoustic lag | sec |
| τ | Shear stress | lb/in ² |
| ϕ | Bank angle | deg |
| ω | Rate of turn | deg/sec or rad/sec |

SUBSCRIPTS

| <u>Symbol or Term</u> | <u>Definition</u> |
|-----------------------|--------------------------------|
| a | Ambient |
| a | Available |
| cr | Critical |
| e | Equivalent |
| ex | Excess |
| f | Final |
| i | Induced |
| i | Initial |
| iw | Corrected to a standard weight |
| L | Laminar |
| M | Wave |
| N | Normal (perpendicular) |
| o | Stagnation or total |
| p | Parasite |

SUBSCRIPTS

| <u>Symbol or Term</u> | <u>Definition</u> |
|-----------------------|-------------------------------------|
| p | Pressure |
| r | Required |
| r | Root |
| s | Static |
| s | Standard day |
| sl | Sea level |
| t | Tangential |
| t | Test day |
| T | Total |
| TD | Touchdown |
| TO | Takeoff |
| X | Conditions upstream of shock wave |
| Y | Conditions downstream of shock wave |
| OL | Zero lift |
| 1,2,3, etc. | Specific condition or station |

SUBSCRIPTS

Symbol or Term

Definition

∞

Free stream condition

SUPERSCRIPT

Symbol or Term

Definition

Choked condition